



# Multi-GNSS for Space Service Volume

Arunkumar Rathinam

PhD Student (ACSER)

Never Stand Still

Faculty of Engineering

Australian Centre for Space Engineering Research (ACSER)



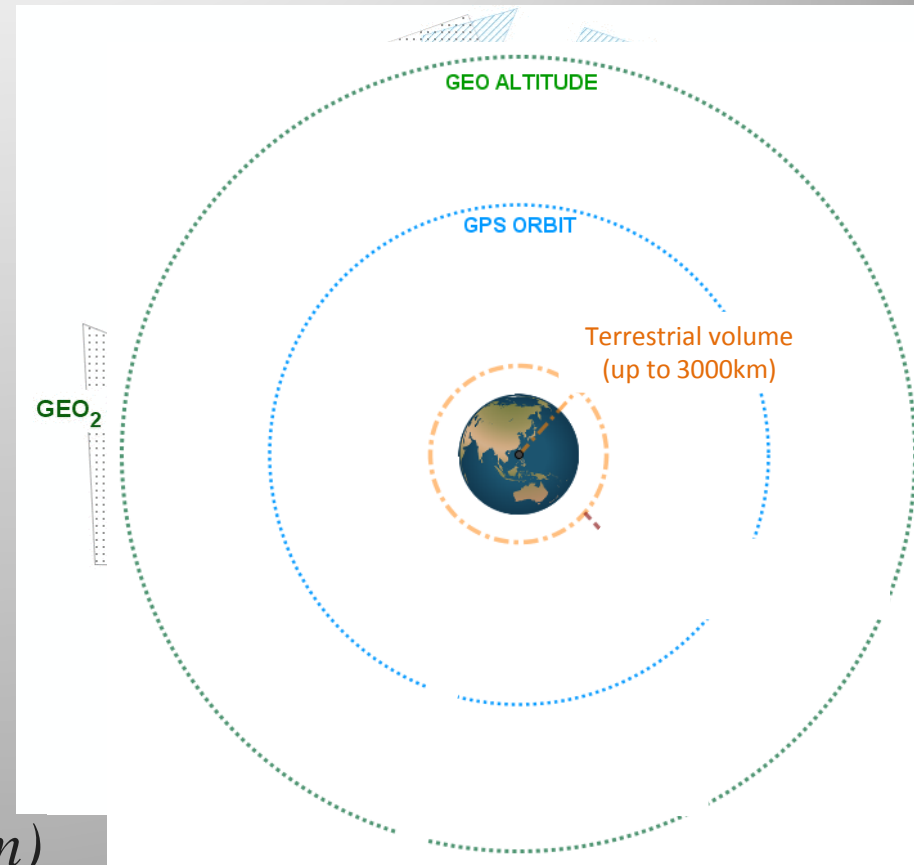
# GNSS - Service regions

- **Terrestrial Service Volume**

- *Earth Surface*
- *LEO (up to 3000 km)*

- **Space Service Volume**

- *MEO (3000 ~ 8000 km)*
  - direct access to GNSS
- *HEO/GEO (8000 ~ 36000km)*
  - depends on GNSS signals available over Earth's limb



# Positioning significance in SSV

**Primary motivation:** *Satellite autonomy*

	<i>Current</i>	<i>Achievable</i>
• Real-time navigation performance	km	meter
• Trajectory manoeuvre recovery	5-10 hrs	minutes
• Reduce expensive on-board clocks	\$100sK - 1M	\$15K - 50K
• Mission operations costs	\$500 - 750K/yr	(savings)

Source: [www.gps.gov/governance/advisory/meetings/2016-05/parker.pdf](http://www.gps.gov/governance/advisory/meetings/2016-05/parker.pdf)

# Multi-GNSS

- **Global Navigation Systems**
  - *GPS (US), GLONASS (Russia)*
  - *Galileo (Europe), BEIDOU (China)*
- **Regional Navigation/Support Systems**
  - *QZSS (Japan)*
  - *NavIC (India)*
- **SBAS (Augmentation Systems)**
  - *EGNOS (Europe), WAAS (US), MSAS (Japan), GAGAN (India), SDCM (Russia)*
- **110 satellites in orbit (at present)**

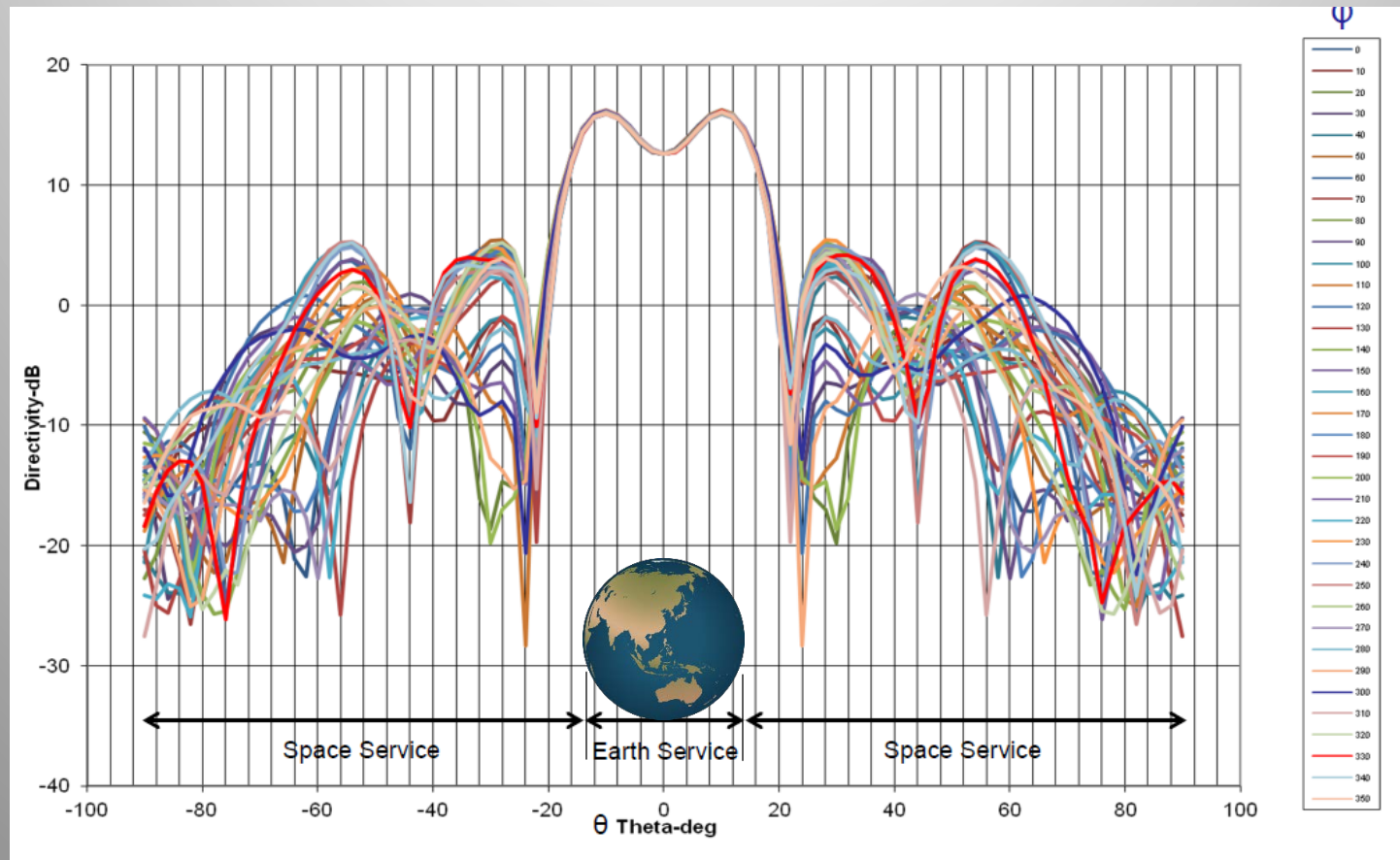
# SSV Positioning - Performance

- **Primary parameters**
  - *Minimum received power level*
    - dominated by path loss
  - *Signal availability*
    - transmitting pattern and orbital characteristics
  - *Pseudorange accuracy*
    - Phase center
    - Non-availability of phase variations (at larger angles)
- **Secondary parameters**
  - *Dilution of precision*

# Received signal power level

	Signal	Ref. off boresight angle (Main) (Degrees)	Min. Received Power (0 dBi RCP ant. at GEO) dBW	URE
<b>GPS</b>	L1	23.5°	-184	~1.1 m
	L2	26°	-183	
	L5	26°	-182	
<b>Glonass</b>	L1	20°	-185	~1.4 m
	L2	28°	-184.4	
	L3	28°	-184	
<b>Galileo</b>	E1	20.5°	-182.5	~1.1 m
	E5	23.5°	-182.5	
	E6	21.5°	-182.5	
<b>Beidou</b>	B1	24° & 18°	-184.1 & -185.8	~1.4 m
	B2	26° & 21°	-182.7 & -184.4	
	B3	26° & 21°	-184.5 & -186.2	
<b>QZSS</b>	L1	22°	-185.3	~1.3 m
	L2	24°	-188.7	
	L5	24°	-180.7	

# Signal availability - Antenna Pattern

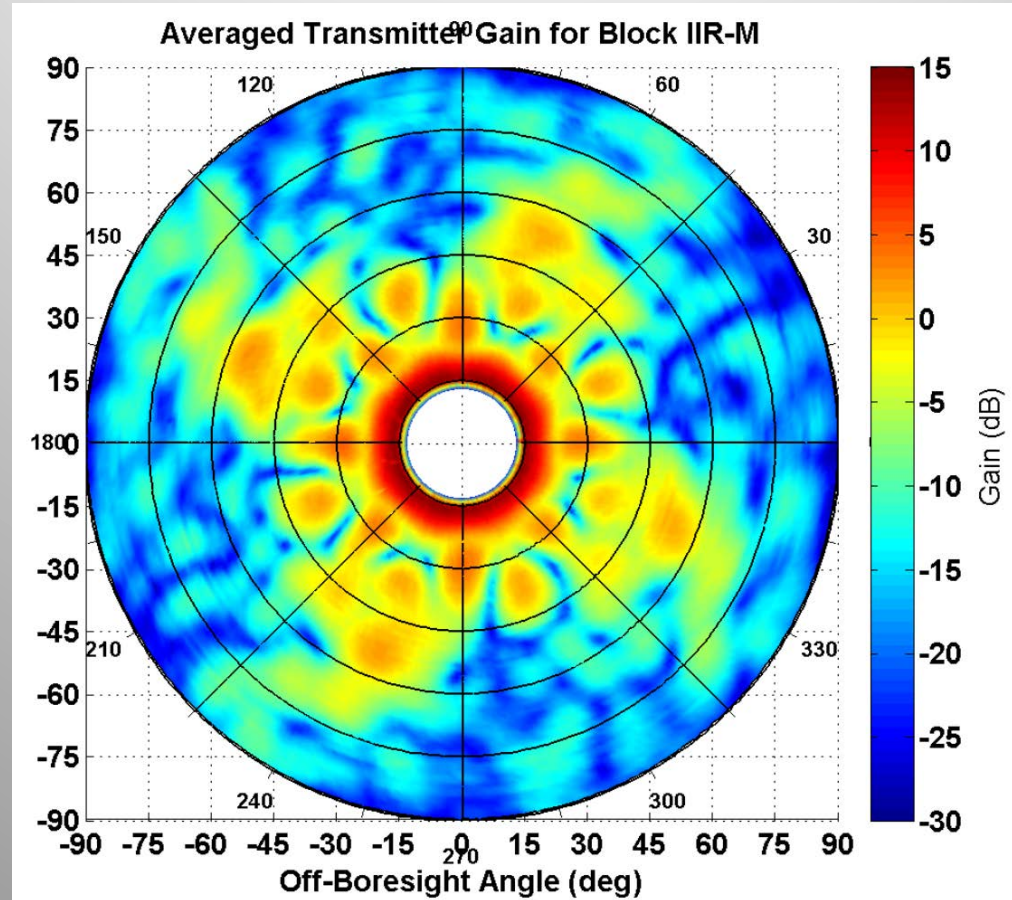


GPS Block IIR and IIR-M Legacy Antenna Pattern L1

© Lockheed Martin



# Signal availability - Antenna Pattern

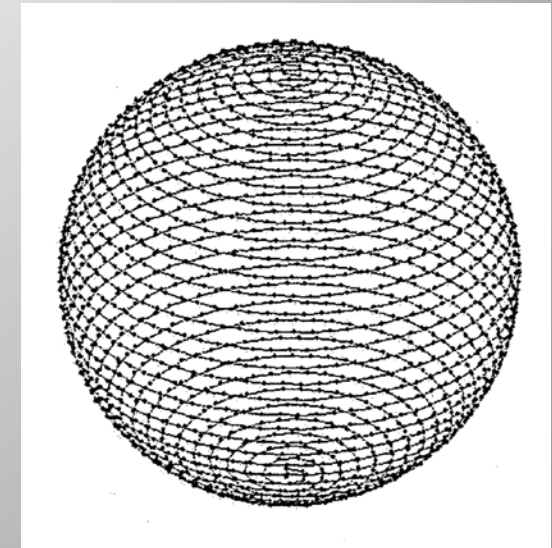


Source: Martzen, Phillip, et al. "GPS Antenna Characterization Experiment (ACE): Receiver Design and Initial Results." (2015).



# SSV simulation – past

- Based on GPS constellation
- Gridded points
  - approx.  $6^\circ$  equatorial grid spacing
- Specific altitudes(in steps)
  - data are interpolated for middle region
- defined time periods
  - simulated for 24 hours
  - not enough to cover the synchronous with GNSS orbital period
- side lobe signals were not simulated



Bauer, F. H., et al. "The GPS space service volume." (2006).

# SSV simulation

- **Line of Sight Visibility**

- *fn (Synodic Period, Receiver ant. beam width, Transmitting ant. beam width, Signal Power)*

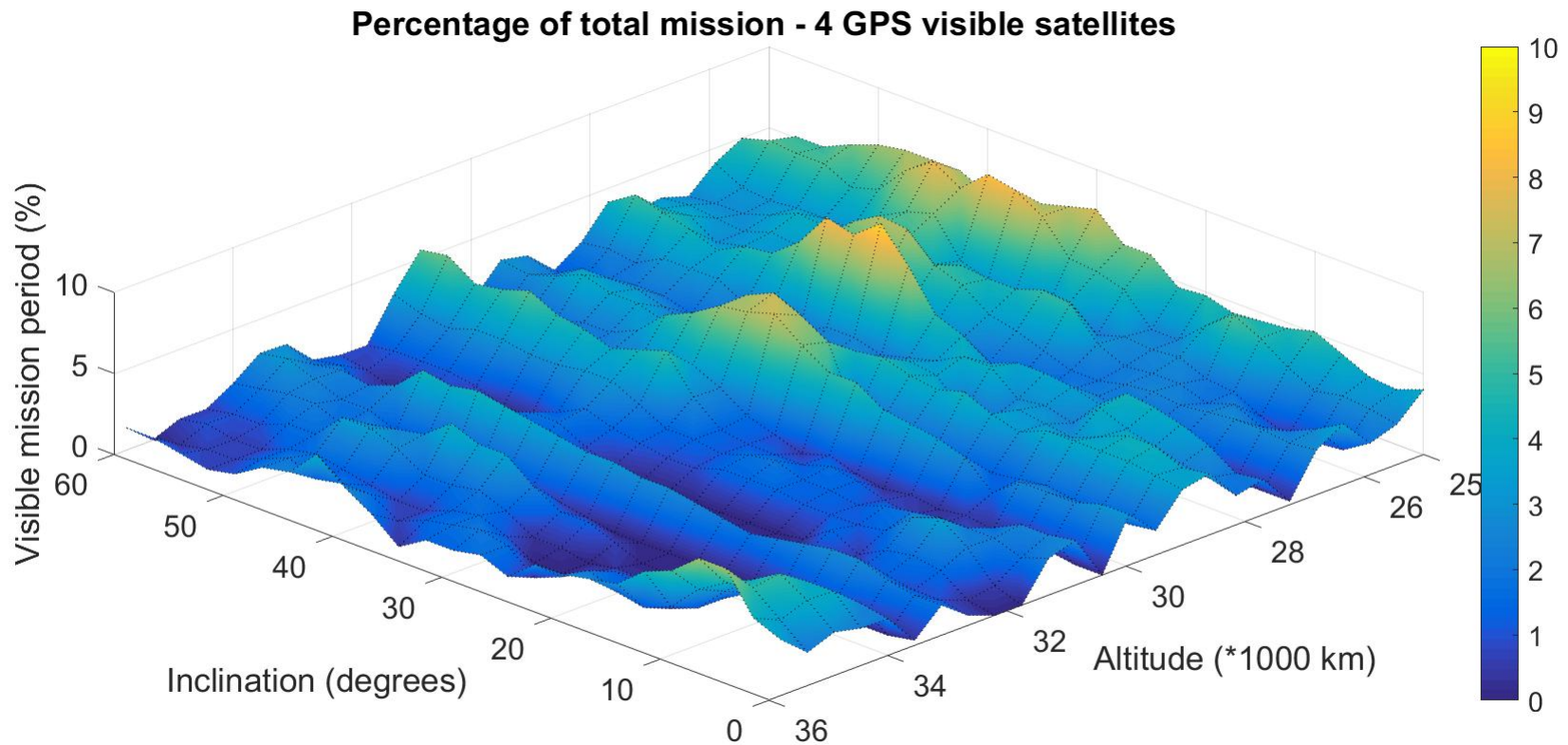
- **Altitude vs Inclination vs Mission duration**

- *Mainly for HEO/GEO SSV*

- Altitudes varies from 25,000 ~ 35,800 km (every 450 km)
    - Inclinations varies from 0° ~ 60° (every 2.5°)
    - 25 x 25 array (Each point corresponding one synodic period)

# GPS availability

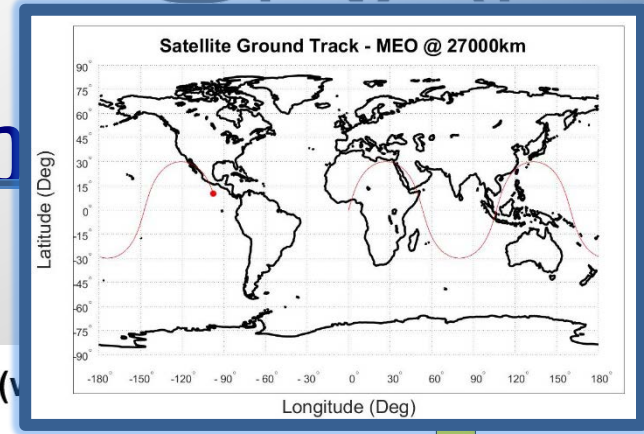
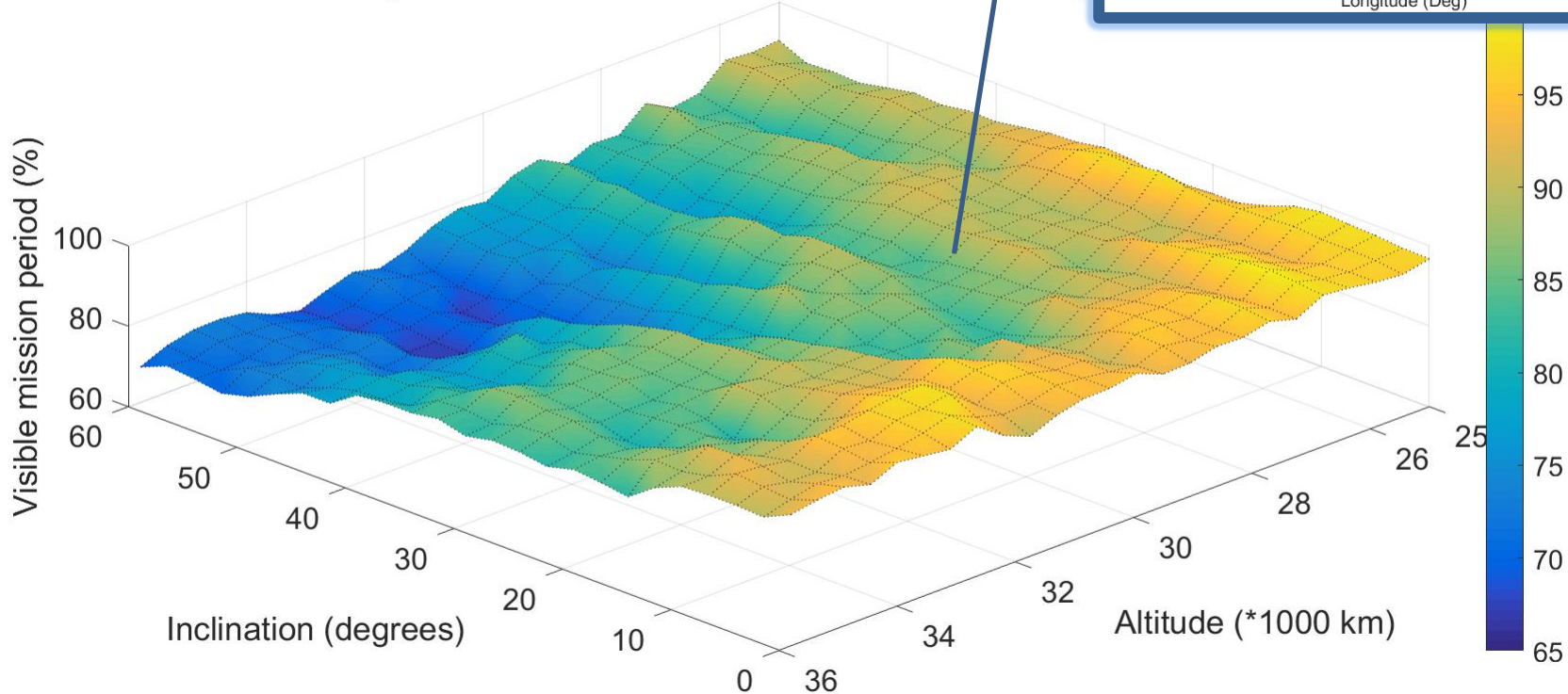
Only Main lobe considered for this simulation ( $23.5^\circ$  L1)



# GPS availability w/side lobe

Side lobe signals ( $28^\circ - 36^\circ$ )

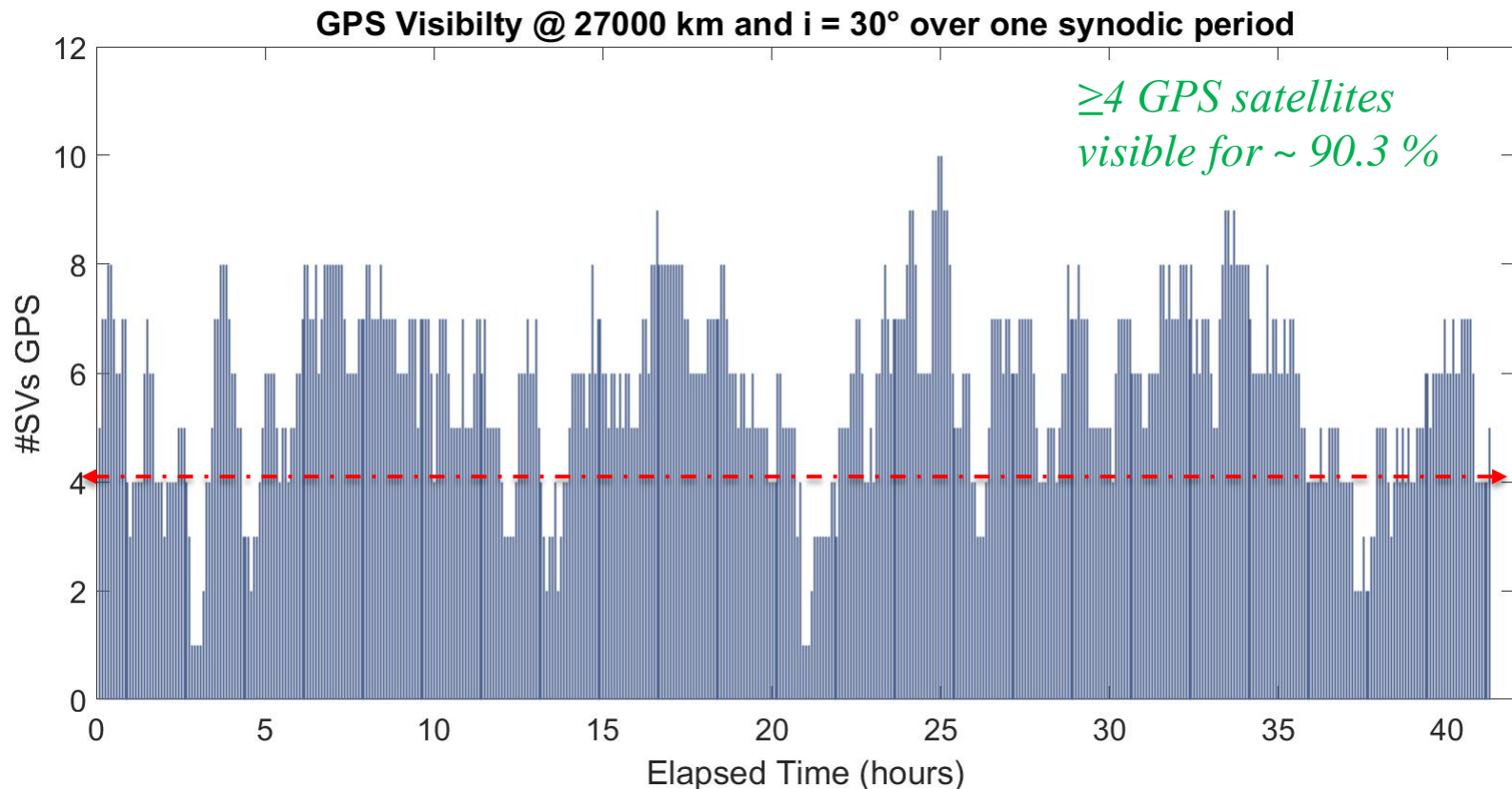
Percentage of total mission - 4 GPS visible satellites (v)



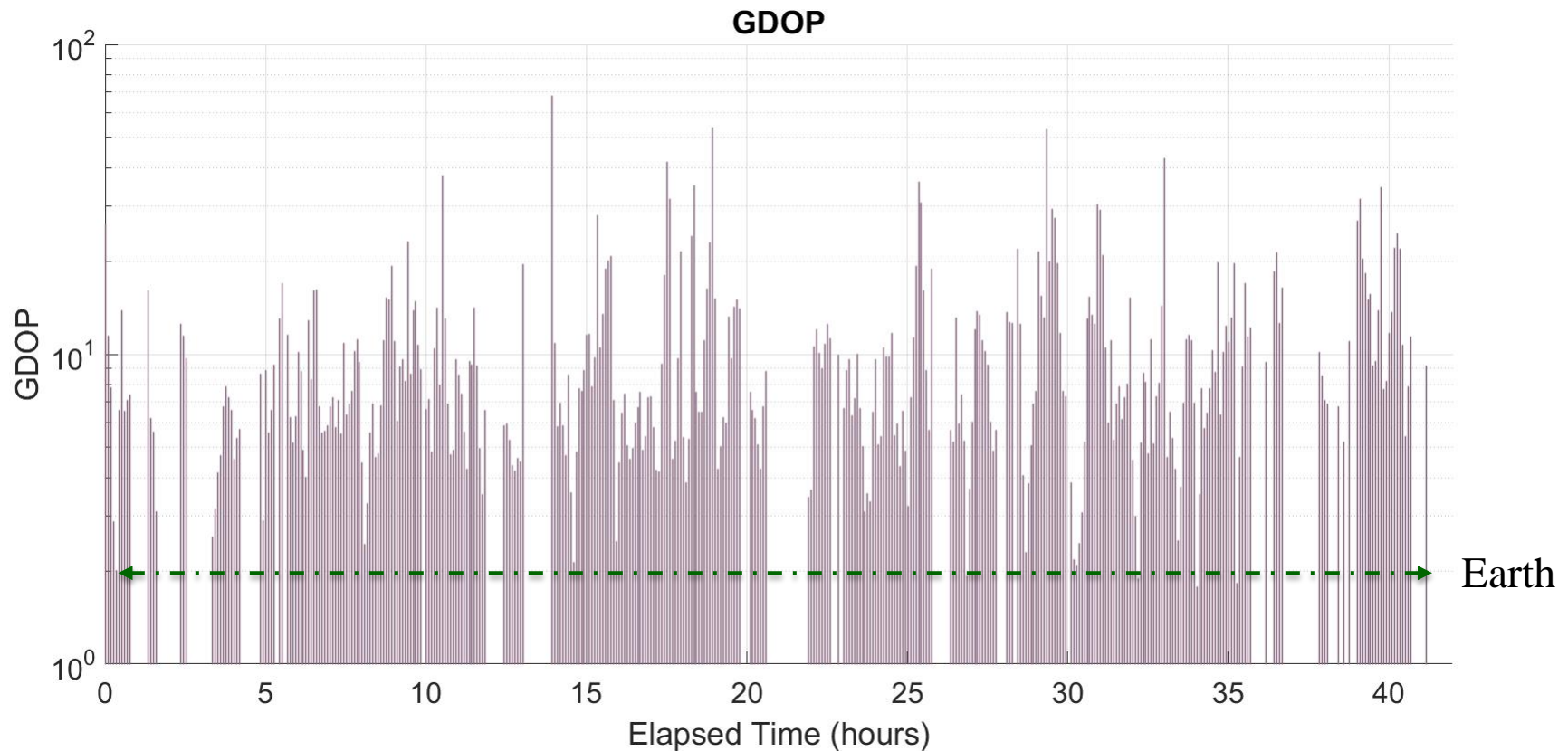


# GPS availability w/side lobes

No of visible satellites recorded at 5-minute resolution



# GPS availability & GDOP





# Multi GNSS – Simulation

- **Design Trade-off**

- *Dual antenna design - receiver*

- Zenith pointed - isotropic antenna
    - Nadir pointed - directional gain antenna

- *Dual frequency receivers*

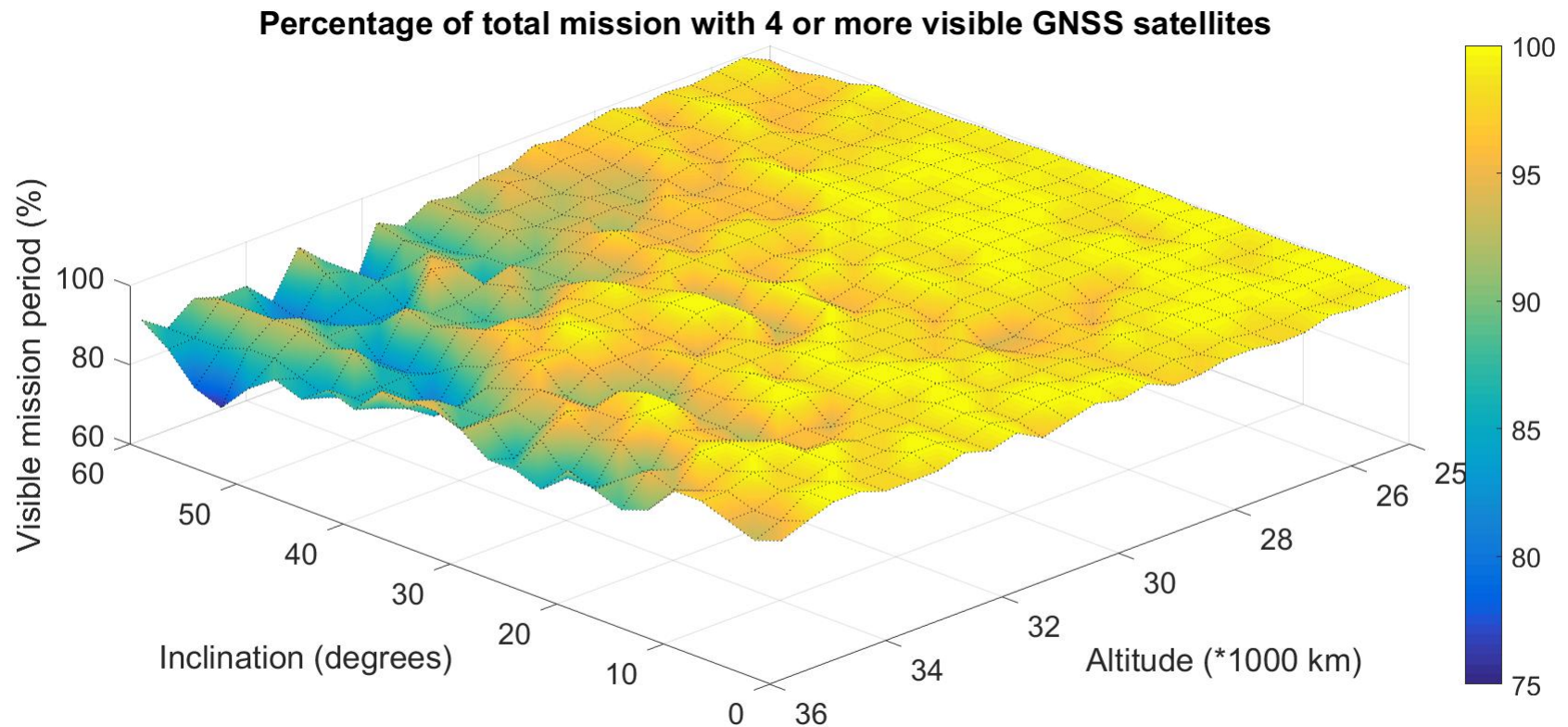
- eliminates ionospheric mask requirement

- *Different Constellation combinations*

- GPS, GLONASS, Galileo, Beidou, QZSS, SBAS
    - Theoretically preferred : constellations at higher altitudes and larger beam width antennas

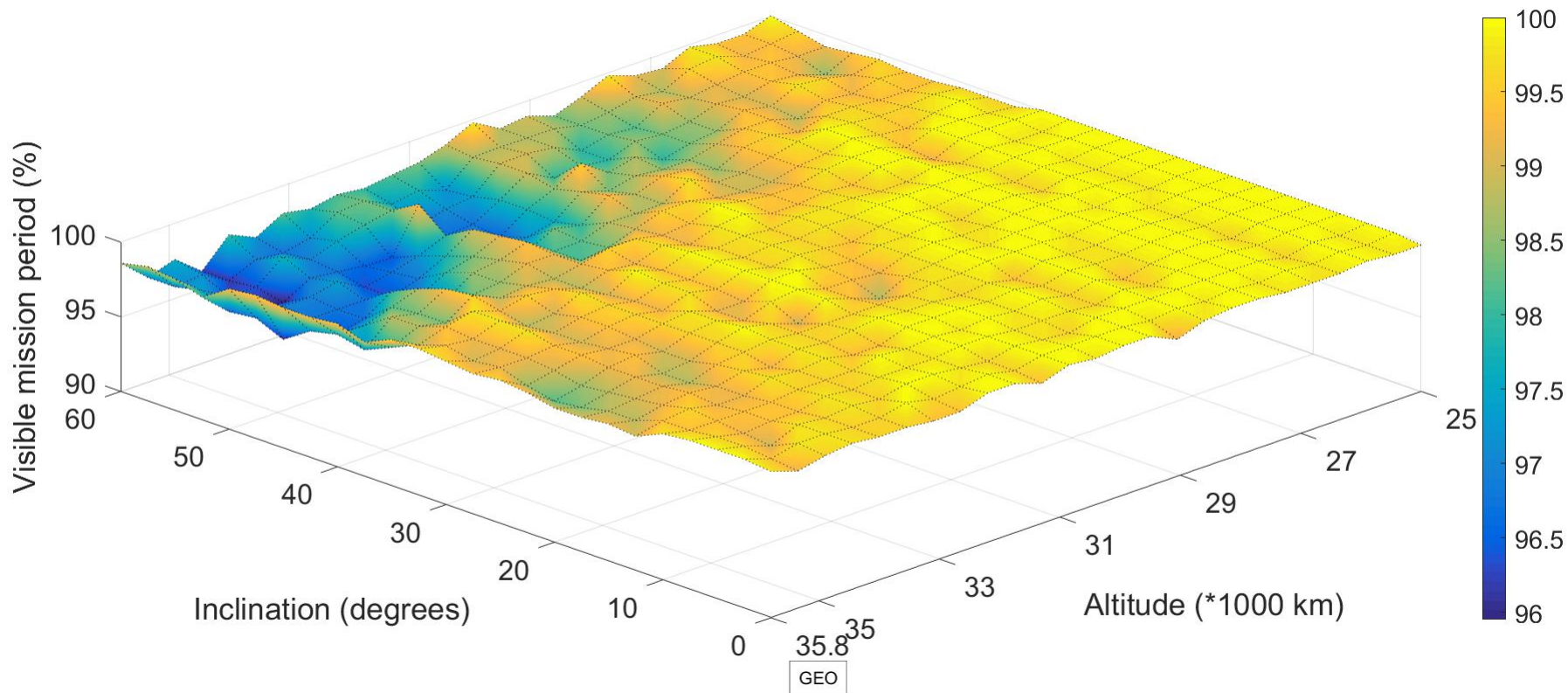
# Multi GNSS – Results (1)

Minimum number of systems – GPS + GLONASS + QZSS + SBAS



# Multi GNSS – Results (2)

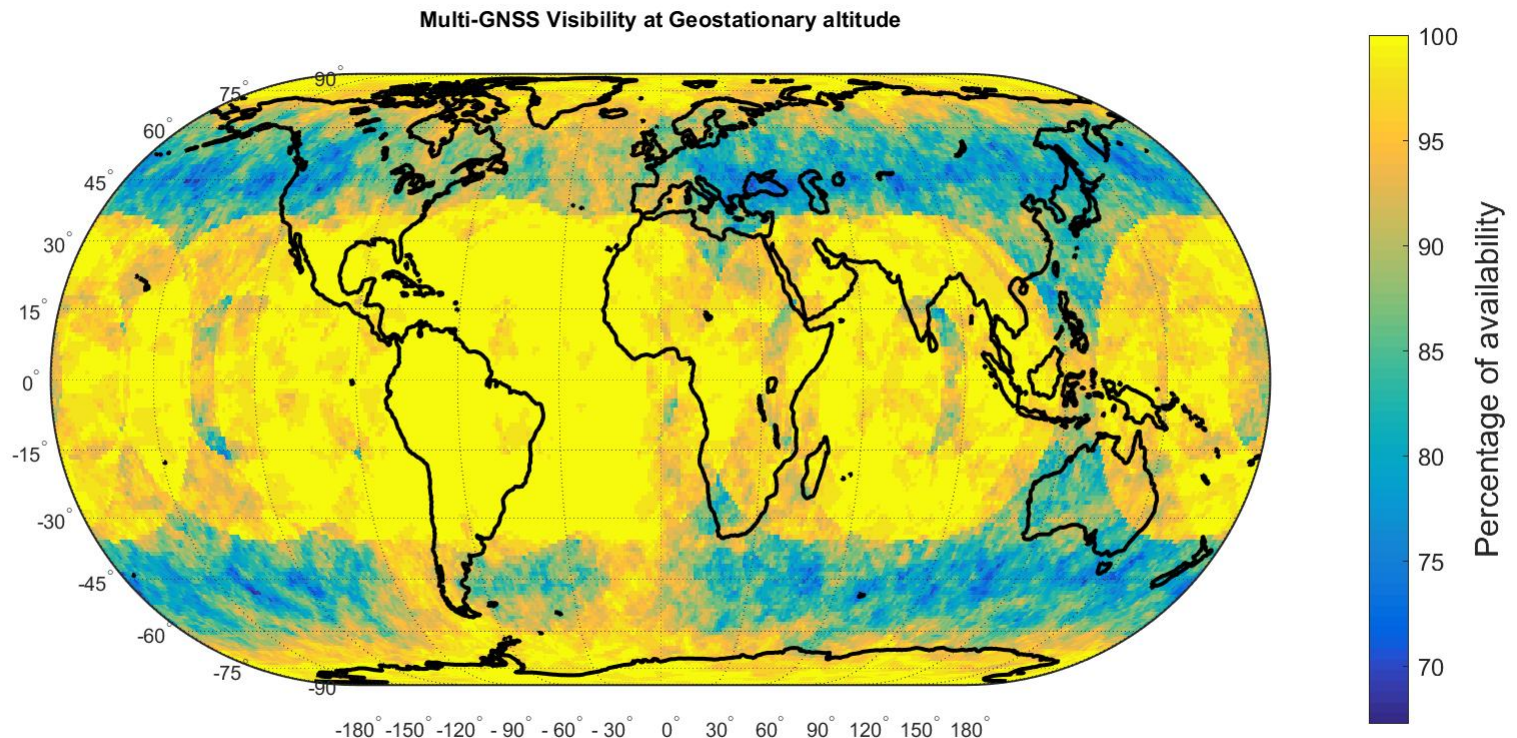
GPS + GLONASS + GALILEO + QZSS + SBAS





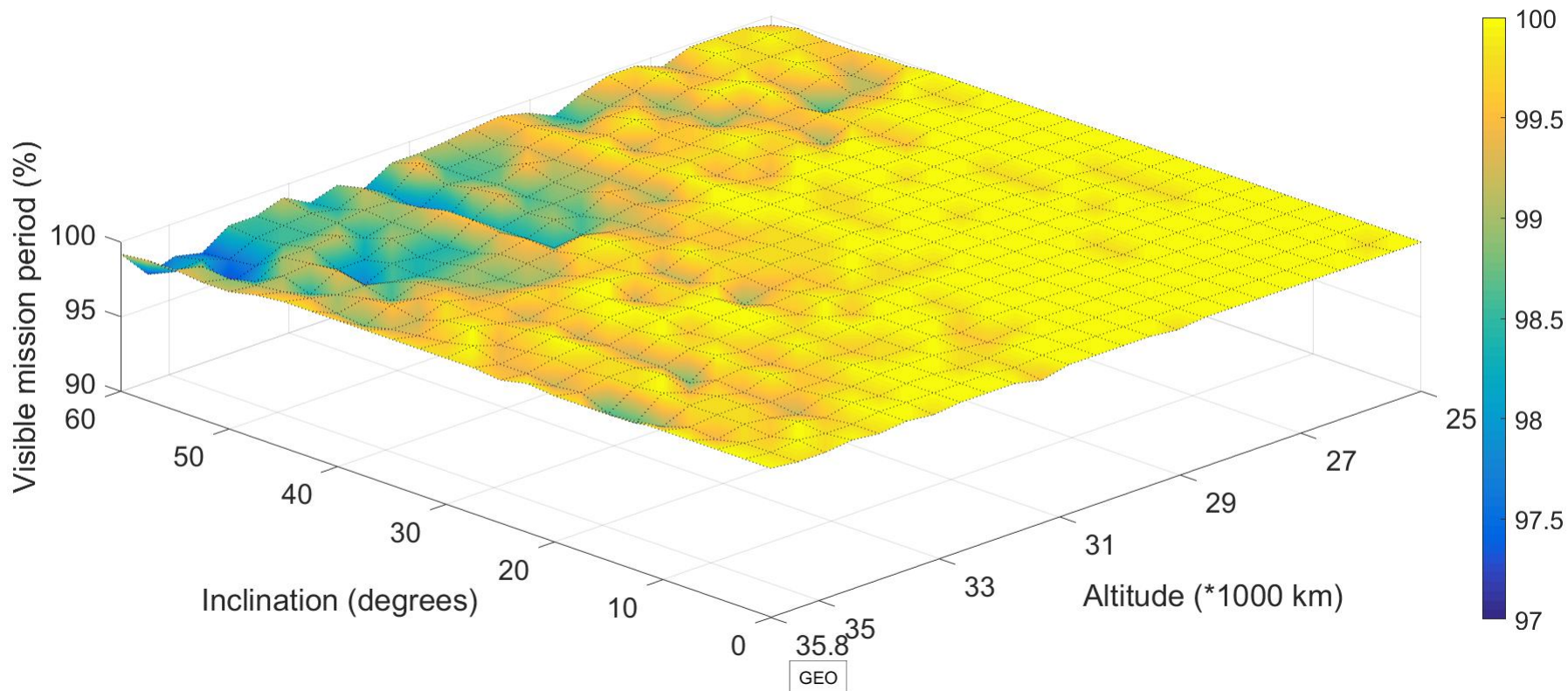
# Multi GNSS – Results (3)

GPS + GLONASS + GALILEO + QZSS + SBAS (@ GEO altitude)



# Multi GNSS – Results (4)

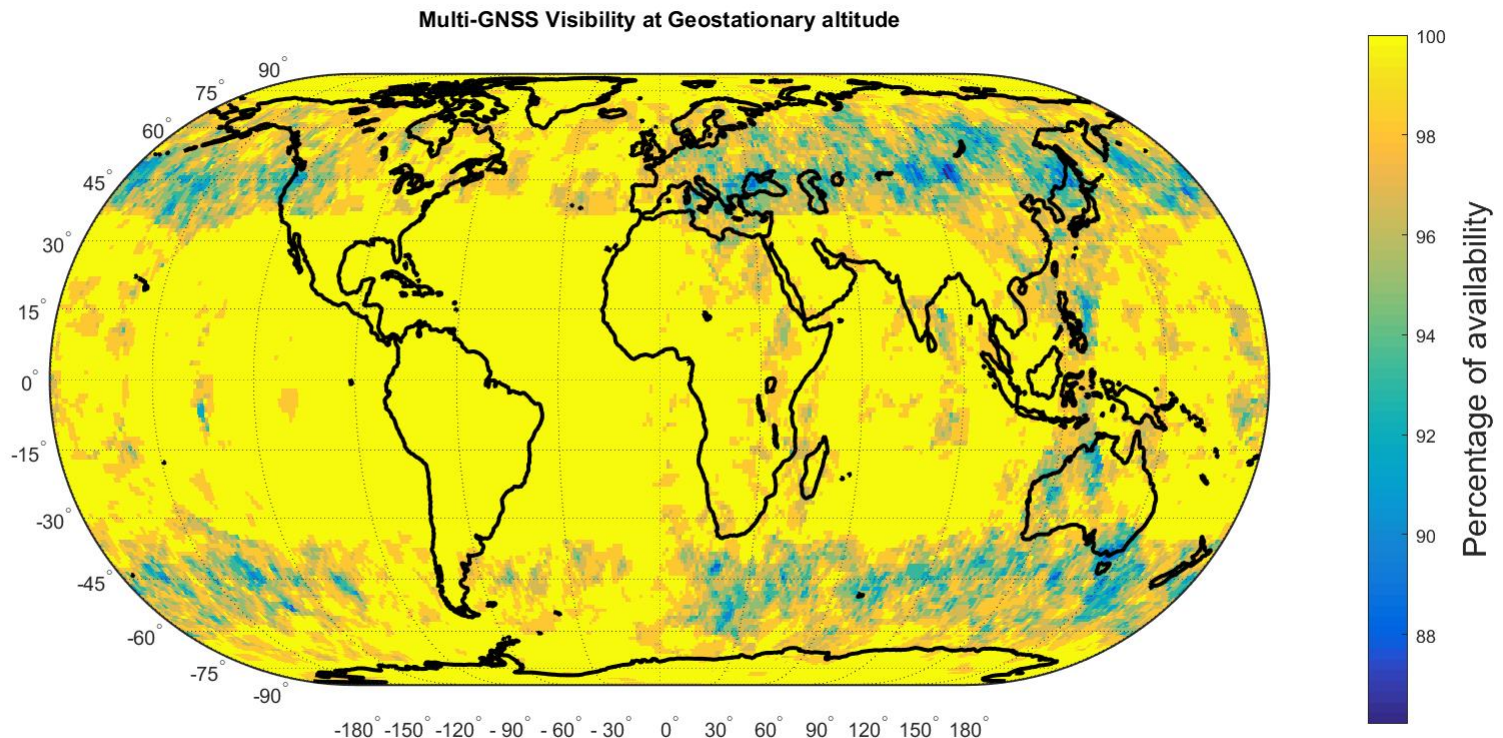
GPS + GLONASS + BEIDOU + QZSS + SBAS





# Multi GNSS – Results (5)

GPS + GLONASS + BEIDOU + QZSS + SBAS (@ GEO altitude)





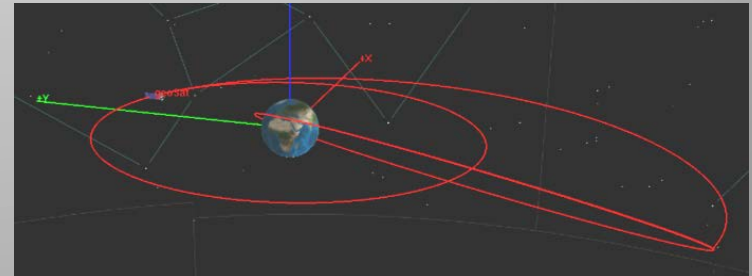
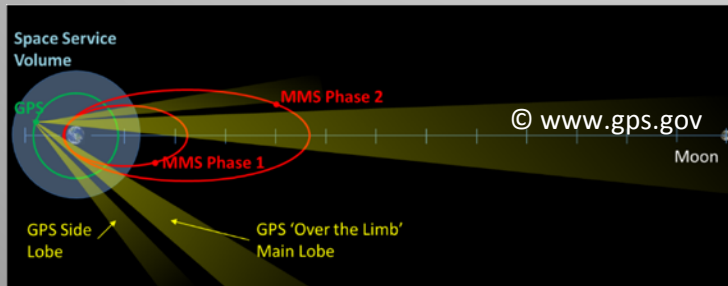
# Conclusion

- Combined GNSS offers **better options** for positioning in higher altitudes
- For best solution
  - Careful selection of **GNSS constellation and frequency combinations are necessary** based on mission scenario
- GNSS Side Lobe Signals specification is **critical**
  - Not only increases availability for positioning in SSV
  - Opens wide array of opportunities for low cost missions

# Conclusion

- **Future**

- Extended the Scope to Even higher orbits such as 76,000 km altitudes
- Specifically for mission scenarios
  - such as formation flying & Geo-Transfer Orbit Scenarios



- Possible state estimation process with  $<4$  satellites available

THANKS !

# GNSS space receivers (ESA missions)

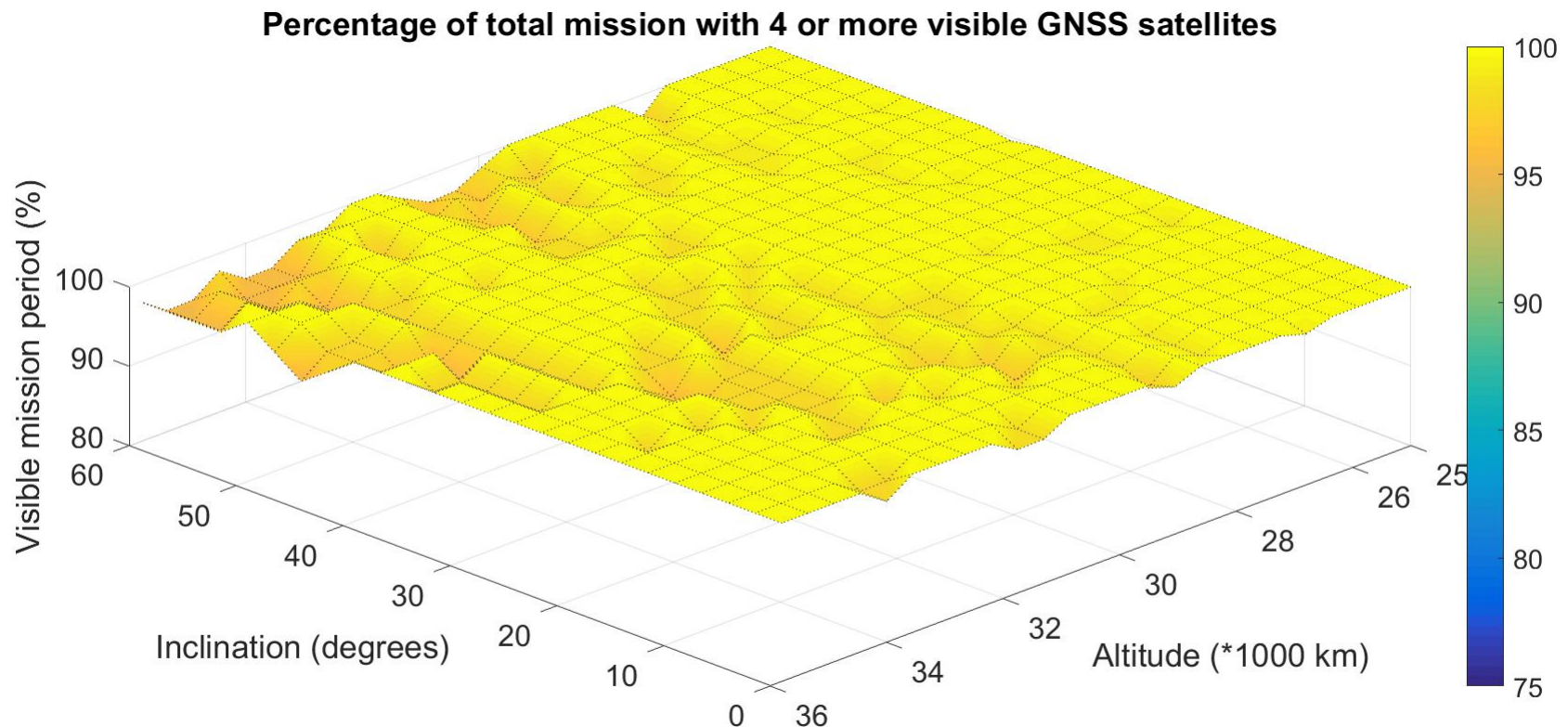
Application		Mission Examples	Orbit
<b>Absolute Navigation (Platform Rx)</b>	LEO Orbit	PLEIADES, UK-DMC, all ESA Earth Obs. (Explorers, Copernicus, MetOp), GlobalStar, Proba-2, Demeter, TDS-1	LEO
	GEO/GTO/HEO Orbit	STENTOR, SkyLAN, IntelSat, GMP, SmallGEO, MTG, STE-QUEST, Proba-3	GEO/HEO
	Re-entry	ARD, Pre-X	LEO to ground
	Launcher	Ariane V, VEGA	Ground to GTO
<b>Relative Navigation (Platform Rx)</b>	Rendezvous	ATV	LEO
	Formation Flying	GRACE, PRISMA, Proba-2, MMS, TerraSAR-x, FF Xeus, NGGM	LEO/HEO GEO
<b>EO/Scientific Instruments</b>	Support to POD	ESA Explorers (GOCE, SWARM, ...), Copernicus Sentinels, MetOp-SG, MetOp, CHAMP, GRACE	LEO
	Atm. Sounding	MetOp, CHAMP, Jason-CS, MetOp-SG	LEO HEO
	Reflectometry	UK-DMC, TDS-1, GEROS-ISS	LEO HEO
<b>Support to Other Systems</b>	Attitude	PLEIADES, ROCSAT, ALPHABUS	LEO GEO
	Timing	GEO telecom, GlobalStar, Iridium, MTG	LEO GEO

# SSV Requirements (based on GPS)

- Current requirement
  - Signal availability
  - Minimum received signal power at GEO
  - Maximum pseudo range accuracy
- Proposed requirement (adds second tier of capability for HEO/GEO users)
  - Increased signal availability to nearly continuous for at least 1 signal
  - Relaxed pseudo range accuracy from 0.8m RMS to 4m RMS
  - No change to minimum received signal power
  - Applies to all signals (L1/L2/L5), all codes

# Multi-GNSS

GPS (main + side) + GLONASS + GALILEO + BEIDOU + QZSS





# Multi GNSS – Results (2)

Minimum number of systems – GPS + GALLILEO + QZSS + SBAS

